

# HEAT TRANSFER FLOW IN A CHANNEL BOUNDED BY STRETCHING WALLS WITH HALL CURRENTS, THERMAL RADIATION, DISSIPATION AND NON-UNIFORM HEAT SOURCES

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**Abstract:** We discuss the effect of thermal radiation and Hall currents on convective heat transfer flow of an electrically conducting, viscous fluid in a vertical channel bounded by stretching walls in the presence of non-uniform heat sources. The non-linear equations have been solved by Runge-Kutta method along Shooting technique. It is found that an increase in Hall Parameter( $m$ ) reduces the skin friction and enhances the Nusselt number at the walls. Higher the thermal radiation smaller  $Nu$  at the left wall and larger at the right wall. An increase in  $Ec$  increases  $Nu$  at  $\eta = \pm 1$ .

**Keywords:** Thermal Radiation, Hall Effects, Dissipation, Non-Uniform Heat Sources, Stretching Walls.

**1. Introduction:** Laminar flows through channels have applications in the fields of gas diffusion, ablation cooling, filtration, microfluidic devices, surface sublimation, grain regression (as in the case of combustion in rocket motors) and the modelling of air circulation in the respiratory system. Laminar air-flow systems have been used by the aerospace industry to control particular contamination. The laminar flow cabinet have been used in the maintenance of negative pressure and in the adjustment of the fans to exhaust more air. Therefore, the Navier-Stokes equations which are the governing equations for these problems have attracted the interest of the researchers. Sutton and Barto(16) described an exact solution of Navier-Stokes equations for motion of an incompressible viscous fluid in a channel with different pressure gradients. Their solutions are helpful in verifying and validating computational models of complex unsteady motions, to guide the design of fuel injectors and controlled experiments. Simulation of flow through microchannels with design roughness was presented numerically by Rawool et al(11). A numerical investigation is made by Robinson(12) for the problem of steady laminar incompressible flow in a porous channel with uniform suction at both walls. Taylor et al(17) studied three dimensional flow by uniform suction through parallel porous walls. The investigations of Taylor(17) were further extended to a more general three dimensional stagnation point which can capture the phenomena in a single class of state by Hewitt et al(7). Two dimensional viscous incompressible fluid flow between two porous walls with uniform suction was analysed by Cox(4). Berman(2) proposed the two dimensional laminar steady flow through a porous channel which was driven by suction or injection. Similarity one/two dimensional laminar flow in a porous channel with wall suction or injection was examined analytically by Laurent et al(10). The problem of fluid flow in a channel with porous walls was solved by Karode(8). The exact solution for two dimensional steady laminar flow through a porous channel was generalized by Terril(18), Shresthal and Terril(14,15), Brady(3), Waston et al(20) and Cox(4) under varied conditions. Deng and Martinez(5) worked on two dimensional flow of a viscous fluid in a channel partially filled with porous medium with wall suction. Wang(19) worked on viscous flow due to stretching sheet with slip and suction and proved a closed form unique solution for two dimensional flows. For asymmetric stretching both existence and uniqueness were shown. Muhammad Asraf et al(1) investigated micropolar fluid flow in a channel with shrinking walls. Hajipour and Dehkordi(6) studied the transient behaviour of fluid flow and heat transfer in vertical channel partially filled with a porous medium including the effects of inertial term and viscous

dissipation. Kasif Ali et al(9) have discussed numerical study of micropolar fluid flow and heat transfer in a channel with shrinking and stationary walls. Sarojamma et al(13) have analysed the convective heat transfer flow of a Casson fluid in channel bounded by stretching walls.

In this paper, we investigate the effect of thermal radiation and Hall currents on convective heat transfer flow of an electrically conducting, viscous fluid in a vertical channel bounded by stretching walls. The nonlinear governing equations have been solved by using Runge-Kutta shooting technique. The velocity, temperature have been analysed for different variations of  $m, A_1, B_1, Rd, Ec$ . The rate of Skin friction, rate of heat transfer are evaluated numerically for different variations.

2. Formulation of The Problem:

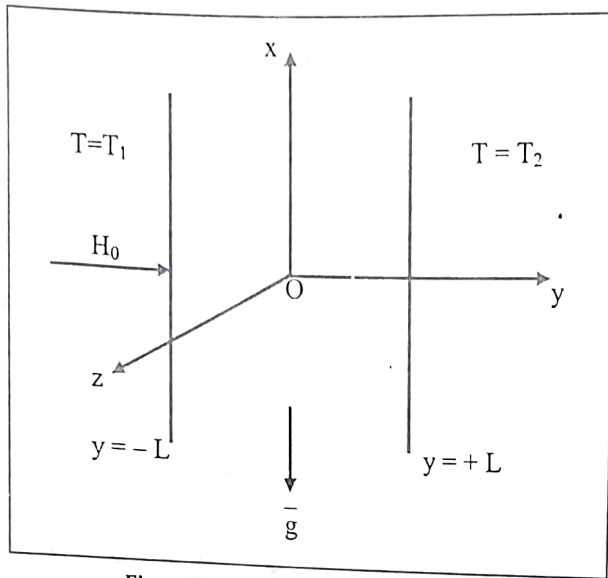


Fig. 1: Configuration of the Problem

The steady two dimensional MHD convective heat and mass transfer flow of a viscous, electrically conducting fluid in a vertical channel bounded by stretching sheets is considered. We consider a rectangular coordinate system  $O(x,y,z)$  with the walls at  $y = \pm L$ . A uniform magnetic field of strength  $H_0$  is applied normal to the walls. Assuming magnetic Reynolds to be small we neglect induced magnetic field in comparison to the applied field. Taking Hall currents into account the equations governing the flow and heat transfer under Boussinesq approximation and Rosseland approximation in the presence of non-uniform heat sources, are

$$u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = \nu \frac{\partial^2 u}{\partial y^2} - \frac{\sigma \mu_e^2 H_0^2}{1+m^2} (u + mw) + \beta g (T - T_0) \quad (2.1)$$

$$u \frac{\partial w}{\partial x} + v \frac{\partial w}{\partial y} = \nu \frac{\partial^2 w}{\partial y^2} - \frac{\sigma \mu_e^2 H_0^2}{1+m^2} (mu - w) \quad (2.2)$$

$$\rho C_p (u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y}) = k_f \frac{\partial^2 T}{\partial y^2} + (\frac{k_f u s}{x \nu}) (A_1 (T_1 - T_2) u + B_1 (T - T_2)) + \frac{16 \sigma^* T_0^3}{3 \beta_R} \frac{\partial^2 T}{\partial y^2} + \frac{\sigma \mu_e^2 H_0^2}{1+m^2} (u^2 + w^2) + 2 \mu ((\frac{\partial u}{\partial y})^2 + (\frac{\partial w}{\partial y})^2) \quad (2.3)$$

where  $A_1$  and  $B_1$  are coefficients of space dependent and temperature dependent internal heat generation or absorption respectively. It is noted that the case  $A_1 > 0$  and  $B_1 > 0$ , corresponds to internal heat generation and that  $A_1 < 0$  and  $B_1 < 0$ , the case corresponds to internal heat absorption case.  $\sigma^*$  is the

Stephan – Boltzmann constant and  $\beta_R$  is the mean absorption coefficient.  $T$ ,  $k_f$ ,  $C_p$ ,  $\beta$ ,  $q_r$ , are defined as temperature of the fluid, thermal diffusivity, specific heat at constant pressure, coefficient of thermal expansion, radiative heat flux respectively.,

The boundary conditions for the velocity, temperature are

$$u(x, -L) = us = bx, u(x, +L) = bx, v(x, \pm L) = 0$$

$$T(x, -L) = T_1, T(x, +L) = T_2 \tag{2.4}$$

Where  $b > 0$  is the stretching rate of the channel wall,  $T_1, T_2$  (with  $T_1 > T_2$ ) are the fixed temperature of the left and right walls respectively.

Introducing the following Similarity variables

$$\eta = \frac{y}{L}, u = bx f'(\eta), v = -bL f(\eta),$$

$$w = bx g_o(x), \theta = \frac{T - T_2}{T_1 - T_2} \tag{2.5}$$

the non-dimensional governing equations are

$$f^{iv} + Re_x (f'''f - ff''') - \frac{M}{1+m^2} (f' + mg) + Gr(\theta) = 0 \tag{2.6}$$

$$g'' + Re_x (fg'_o - fg'_o) - \frac{M}{1+m^2} (mf' - g_o) = 0 \tag{2.7}$$

$$Rd\theta'' + Pr Re_x (f\theta') + Pr(A_{11}f' + B_{11}\theta) + \frac{Pr EcM^2}{1+m^2} (f'^2 + g^2) + Pr Ec (f'')^2 + (g')^2 = 0 \tag{2.8}$$

Where  $Gr = \frac{\beta g(T_1 - T_2)L}{bx}$ , is the Grashof number,  $M^2 = \frac{\sigma \mu_e^2 H_o^2 L^2}{\nu x}$ , magnetic parameter,  $R = \frac{2\Omega}{xL}$ ,

Rotation parameter,  $Pr = \frac{\mu C_p}{k_f}$  is Prandtl number,  $Re_x = \frac{bL^2}{\mu}$ , is the local Reynolds number,

$Ec = \frac{b^2 x^2}{C_p \Delta T}$  is the Eckert number,  $m = \omega_e \tau_e$  is the Hall parameter.

Boundary conditions(2.4), in view of equation(2.5) in non-dimensional form reduces to

$$f(\pm 1) = 0, f'(\pm 1) = 1, g_o(\pm 1) = 0, \theta(-1) = 1, \theta(+1) = 0 \tag{2.9}$$

**3. Numerical Procedure:** The equations(2.6)-(2.8) have been solved by using a shooting technique is consolidated to solve the above system, which may be a combination of the Runge-Kutta method (to solve first order ODE) and a five dimensional zero discovering algorithm (to locate the missing coordinates). It is note that the missing initial conditions are coupled so the arrangement fulfills the boundary conditions

**4. Results And Discussion:** In this analysis we investigate the impact of thermal radiation and Hall effects on convective heat and mass transfer flow of a viscous, dissipative fluid in a vertical channel bounded by stretching plates in the presence of non-uniform heat sources. The non-linear governing equations have been solved by utilizing Fourth order Runge-Kutta – Shooting technique.

The primary velocity, secondary velocity, temperature and concentration have been analysed for different variations of  $m, A_{11}, B_{11}, Rd, Ec$ . We also study Skin friction components ( $\tau_x, \tau_z$ ), the rate of heat transfer (Nu), briefly.

1. The variation of primary velocity  $f'$  with Hall parameter ( $m$ ) shows that  $f'$  enhances in the region  $(-1.0, 0.5)$  and reduces in the region adjacent to the right wall  $\eta = +1$ . The secondary velocity  $g$  and the temperature enhances with increasing  $m$  in the entire flow region (figs. 2b & 2c). The skin friction component  $\tau_x$  reduces at both the walls with  $m$ . The component  $\tau_z$  enhances at the left wall and reduces at the right wall with increasing  $m$ . An increase in  $m$  enhances with the rate of heat transfer at both the walls (table.1).
  2. Figs. 3a-3c represent the primary, secondary and temperature with space dependent heat source ( $A_{11}$ ). With respect to space dependent heat source /sink parameter ( $A_{11} > 0$  &  $A_{11} < 0$ ) reduces in the region  $(-1.0, 0.5)$  while in the region  $(0.5, 1.0)$ , it enhances with  $A_{11} > 0$  and reduces with  $A_{11} < 0$ . The secondary velocities enhances with  $A_{11} > 0$  and reduces with  $A_{11} < 0$  in the region  $(-1.0, 1.0)$ . The temperature  $\theta$  reduces in the left half and enhances in the right half of the channel with increasing values of  $A_{11} > 0$  while a reversed effect is noticed with that of heat sink (fig. 5c). The skin friction components  $\tau_x$  enhances at  $\eta = -1$  &  $+1$  while an increase in  $A_{11} > 0$  enhances at  $\eta = -1$  and reduces at  $\eta = +1$ . The skin friction  $\tau_z$  reduces with  $A_{11} < 0$  and enhances with  $A_{11} > 0$  at both the walls. An increase in  $A_{11} > 0$  and  $A_{11} < 0$  enhances the rate of heat transfer at  $\eta = \pm 1$  (table.1)
  3. Figs. 4a-4c depict the variation of primary, secondary velocities and temperature with temperature dependent heat sources. In the presence of generating heat sources, the primary velocity reduces in the region  $(-1.0, 0.0)$  and enhances in the region  $(0.0, 1.0)$  and in the case of heat absorbing case we notice a reversed effect in the behaviour of  $f'$  in the flow region. The secondary velocity  $g$  decreases in the flow region  $(-1.0, 0.0)$  and enhances in the region  $(0.0, 1.0)$  with increase in  $B_{11} > 0$  while a reversed behaviour is observed in the case of  $B_{11} < 0$  (fig. 4b). In the presence of heat generating sources, heat is generated which results in an enhancement in the temperature in the entire flow region while the temperature is reduced due to the absorption of energy in the case of heat absorption sources (fig. 4c). The skin friction component  $\tau_x$  reduces at  $\eta = -1$  and enhances at  $\eta = +1$  with increasing  $B_{11} < 0$  and reversed effect is noticed in the behaviour of  $\tau_x$  at the walls.  $\tau_z$  reduces with  $B_{11} < 0$  and enhances with  $B_{11} > 0$  at  $\eta = \pm 1$ . The rate of heat transfer enhances at the left wall and reduces at the left wall with  $B_{11} < 0$  and enhances at both the walls with  $B_{11} > 0$ .
  4. From the figures (5a-5c) we observed that the primary velocity increases in the flow regions  $(-1.0, 0.5)$  &  $(0.0, 0.5)$  and enhances in the regions  $(-0.5, 0)$  &  $(0.5, 1.0)$  with increasing  $Rd$  (fig. 5a). The secondary velocity enhances in the entire flow region with increasing  $Rd$  (fig. 5b). The temperature decays in the region  $(-1.0, -0.5)$  and grows in the region  $(-0.5, 1.0)$  with an increase in radiation parameter ( $Rd$ ) (fig. 5c). A rise in  $Rd$  enhances  $\tau_x$  at  $\eta = -1$  and reduces at  $\eta = +1$  while  $\tau_z$  grows at  $\eta = \pm 1$  with increase in  $Rd$ . Also higher the thermal radiation larger  $Nu$  at the left wall and larger at the right wall (table.1).
  5. From the figures (6a-6c), we observed that the primary velocity reduces in the region  $(-1.0, -0.5)$  &  $(0.0, 0.5)$  and enhances in the regions  $(-0.5, 0.0)$  &  $(0.5, 1.0)$  with increasing values of Eckert number ( $Ec$ ). The secondary velocity enhances in the entire flow region of the channel with increasing  $Ec$  (fig. 6b). The temperature enhances with increase in  $Ec$  (fig. 6c). The Shear stress  $\tau_x$  reduces at  $\eta = +1$  and enhances at  $\eta = -1$  whereas  $\tau_z$  increases at both the walls  $\eta = \pm 1$ . The Nusselt number increases with  $Ec$  at  $\eta = -1$  and  $\eta = +1$ . (table 1).
- 5. Conclusions:** We aim at investigating the effect of Hall currents, non-uniform heat sources, thermal radiation on heat transfer flow of viscous fluid in a vertical channel bounded by stretching walls under a uniform magnetic field. The highly non-linear fourth order momentum equation, second order equation of energy have been solved by using Runge-Kutta shooting method. The impact of different physical parameters like Hall parameter ( $m$ ), Heat source parameters  $A_{11}, B_{11}$ , Eckert number ( $Ec$ ) on the flow characteristics have been studied graphically in detail. An increase in Hall Parameter ( $m$ ) reduces the skin friction ( $\tau_x, \tau_z$ ) and enhances the Nusselt number ( $Nu$ ) at the walls. Higher the thermal radiation smaller Nusselt number ( $Nu$ ) at the left wall and larger at the right wall. An increase in  $Ec$  increases  $Nu$  at  $\eta = \pm 1$ .

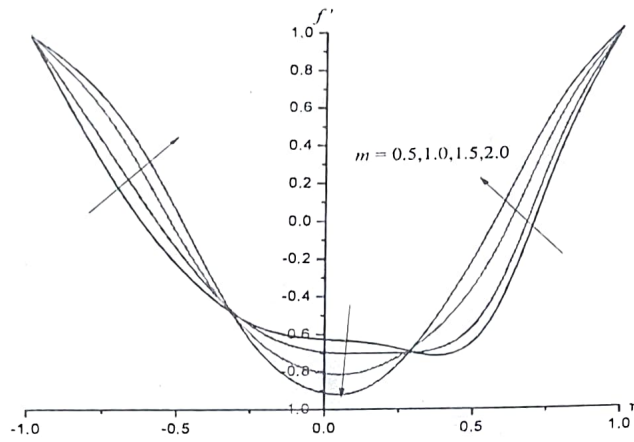


Fig.2a : Variation of  $f'$  with  $m$   
 $m=0.5, Rd=0.5, A_{11}=0.5, B_{11}=0.5, Ec=0.01$

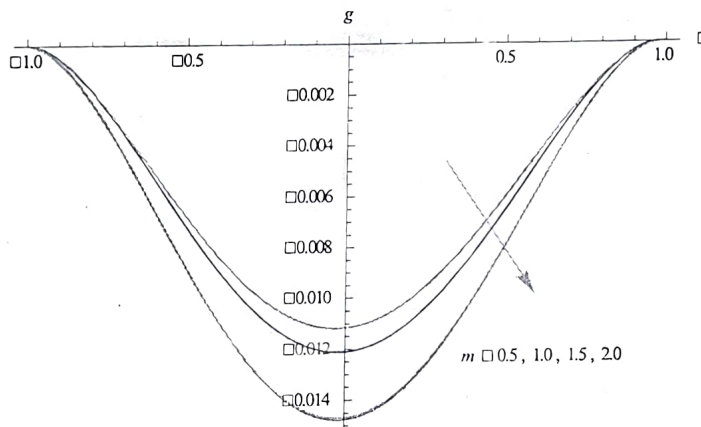


Fig.2b: Variation of  $g$  with  $m$   
 $m=0.5, Rd=0.5, A_{11}=0.5, B_{11}=0.5, Ec=0.01$

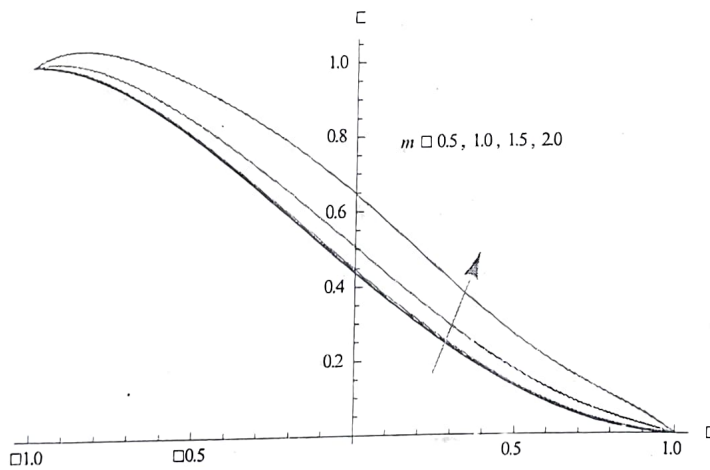


Fig.2c: Variation of  $\theta$  with  $m$   
 $m=0.5, Rd=0.5, A_{11}=0.5, B_{11}=0.5, Ec=0.01$

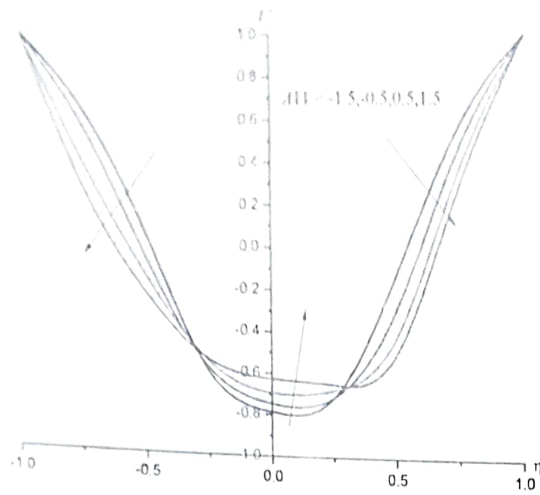


Fig.3a: Variation of  $f$  with  $\Delta_{11}$   
 $m=0.5, Rd=0.5, \Lambda_{11}=0.5, B_{11}=0.5, Ec=0.01$

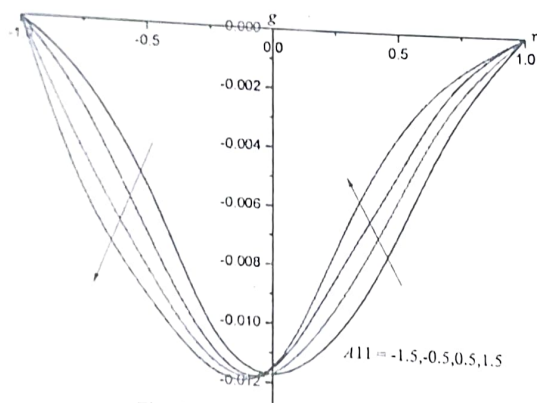


Fig.3b: Variation of  $g$  with  $\Delta_{11}$   
 $m=0.5, Rd=0.5, \Lambda_{11}=0.5, B_{11}=0.5, Ec=0.01$

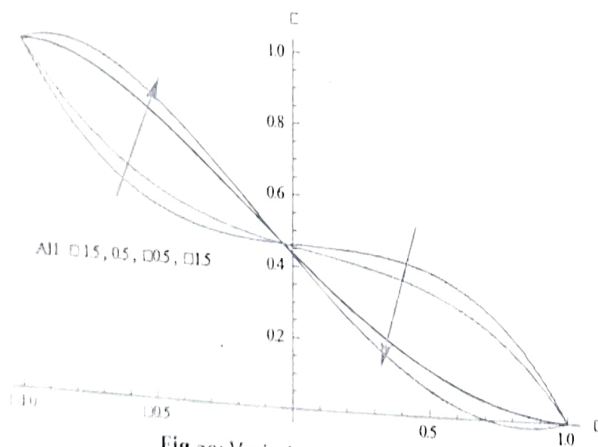


Fig.3c: Variation of  $\theta$  with  $\Delta_{11}$   
 $m=0.5, Rd=0.5, \Lambda_{11}=0.5, B_{11}=0.5, Ec=0.01$

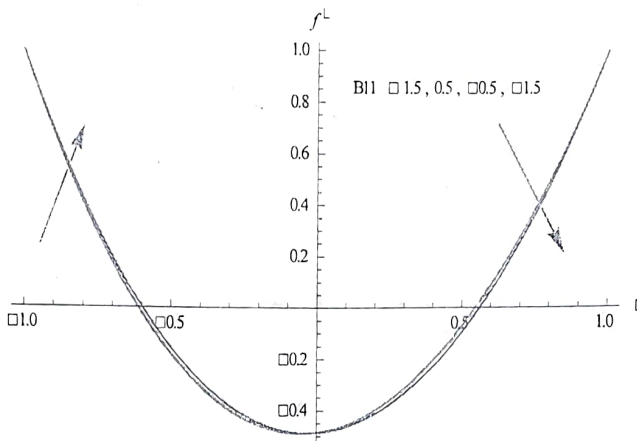


Fig.4a : Variation of  $f^L$  with  $B_{11}$   
 $m=0.5, R_d=0.5, A_{11}=0.5, B_{11}=0.5, E_c=0.01$

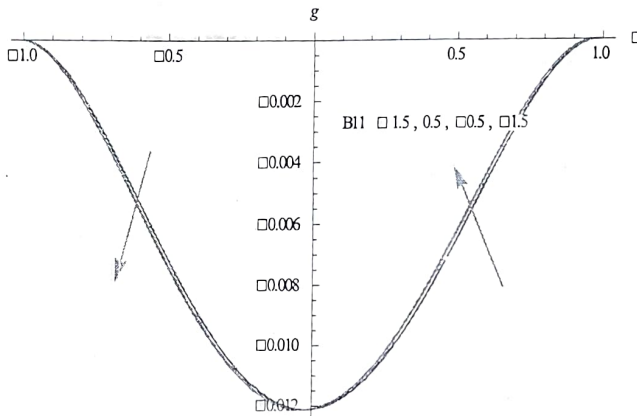


Fig.4b: Variation of  $g$  with  $B_{11}$   
 $m=0.5, R_d=0.5, A_{11}=0.5, B_{11}=0.5, E_c=0.01$

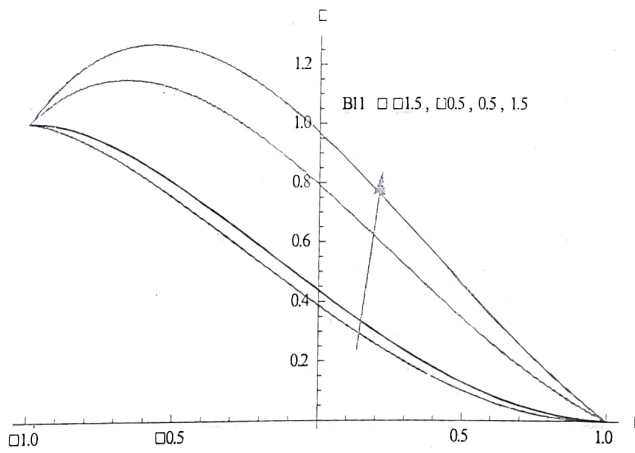


Fig.4c: Variation of  $\theta$  with  $B_{11}$   
 $m=0.5, R_d=0.5, A_{11}=0.5, B_{11}=0.5, E_c=0.01$

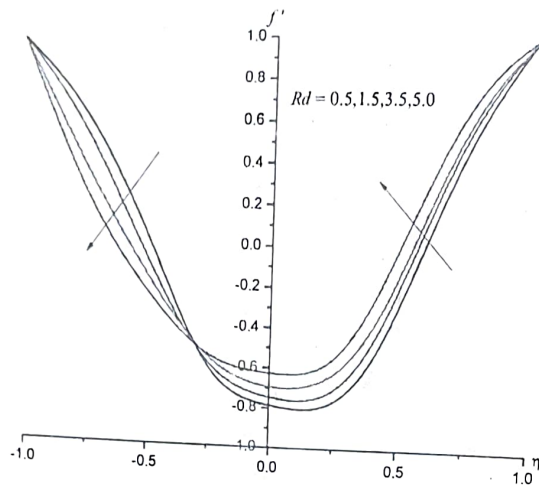


Fig.5a: Variation of  $f''$  with  $Rd$   
 $m=0.5, Rd=0.5, Au=0.5, Bu=0.5, Ec=0.01$

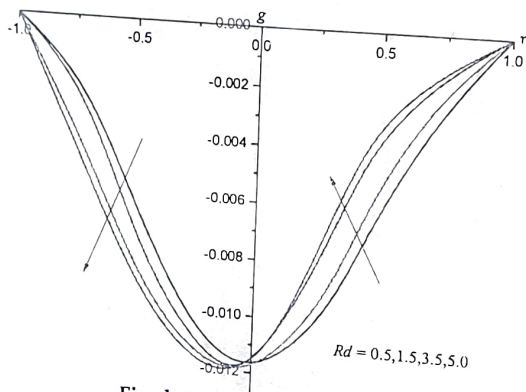


Fig.5b: Variation of  $g$  with  $Rd$   
 $m=0.5, Rd=0.5, Au=0.5, Bu=0.5, Ec=0.01$

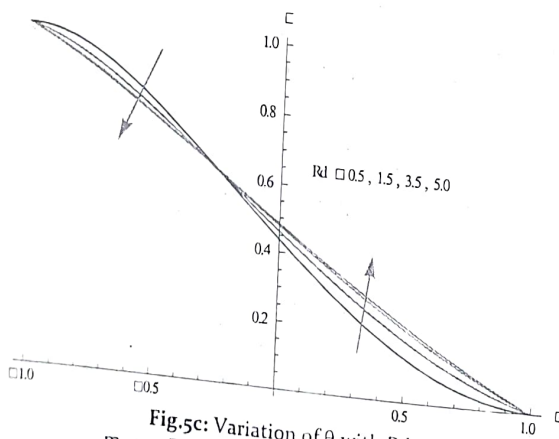


Fig.5c: Variation of  $\theta$  with  $Rd$   
 $m=0.5, Rd=0.5, Au=0.5, Bu=0.5, Ec=0.01$

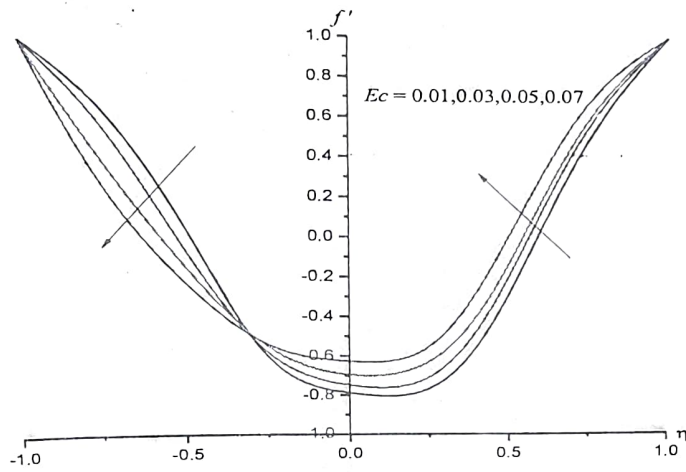


Fig.6a: Variation of  $f'$  with  $E_c$   
 $m=0.5, R_d=0.5, A_{11}=0.5, B_{11}=0.5, E_c=0.01$

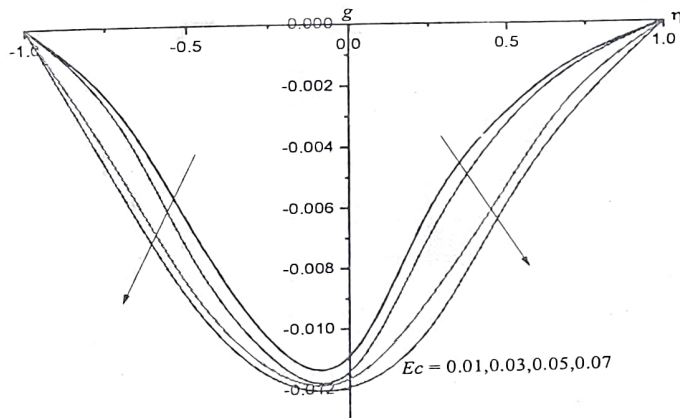


Fig.6b: Variation of  $g$  with  $E_c$   
 $m=0.5, R_d=0.5, A_{11}=0.5, B_{11}=0.5, E_c=0.01$

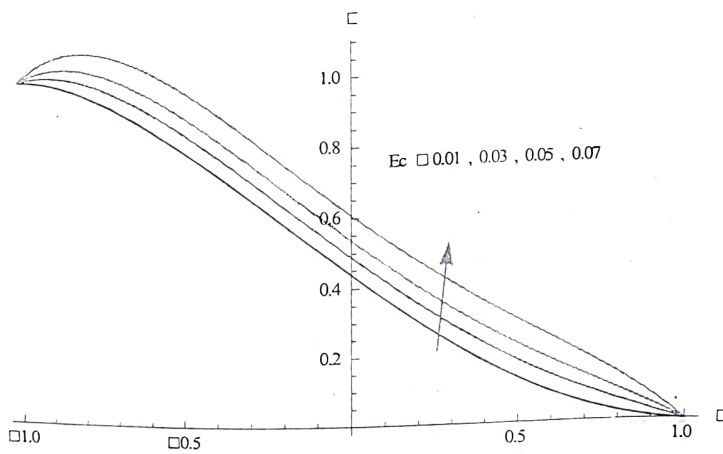


Fig.6c: Variation of  $\theta$  with  $E_c$   
 $m=0.5, R_d=0.5, A_{11}=0.5, B_{11}=0.5$

Table 1: Skin friction ( $\tau_{x,y}$ ), Nusselt number at  $\eta = \pm 1$

Parameter	$\tau_x(-1)$	$\tau_x(+1)$	$\tau_z(-1)$	$\tau_z(+1)$	Nu(-1)	Nu(+1)	
m	0.5	-3.25666	2.95861	-0.000757626	-0.0016044	-0.029767	0.0488547
	1.0	-3.2515	2.94812	-0.00124599	-0.0011163	-0.0545257	0.0808118
	1.5	-3.2508	2.91608	-0.00199101	-0.000555654	-0.229862	0.260591
	2.0	-3.27818	2.87177	-0.00197867	-0.000375453	-0.644577	0.690337
Rd	0.5	-3.25666	2.95861	-0.000757626	-0.0016044	-0.029767	0.0488547
	1.5	-3.25793	2.94973	-0.000795749	-0.00119661	0.209449	0.240197
	3.5	-3.2594	2.94366	-0.000823797	-0.00122341	0.346184	0.359517
	5.0	-3.25991	2.94177	-0.000832658	-0.00123189	0.386331	0.395492
Au	0.5	-3.25666	2.95861	-0.000757626	-0.0016044	-0.029767	0.0488547
	1.5	-3.26043	2.96526	-0.000746207	-0.0011513	-0.312561	-0.240086
	-0.5	-3.23906	2.93373	-0.000788772	-0.00118242	1.13962	1.19851
	-1.5	-3.2348	2.92735	-0.000797643	-0.00118896	1.43006	1.48611
Bu	0.5	-3.25666	2.95861	-0.000757626	-0.0016044	-0.029767	0.0488547
	1.5	-3.24211	2.97084	-0.000664326	-0.00106752	0.145395	-0.00640622
	-0.5	-3.35404	2.87125	-0.00140612	-0.00180257	-0.986485	0.503311
	-1.5	-3.40061	2.82755	-0.00172494	-0.00211625	-1.37771	0.749646
Ec	0.01	-3.25666	2.95861	-0.000757626	-0.0016044	-0.029767	0.0488547
	0.03	-3.27244	2.94281	-0.000864145	-0.00126587	-0.292323	0.291211
	0.05	-3.28727	2.92808	-0.000963983	-0.0013646	-0.540318	0.510564
	0.07	-3.30957	2.90597	-0.0011404	-0.0015128	-0.916583	0.836042

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